

Pilot's Operating Handbook

2. Valid to aircraft S/N:

P1001067

Pilot's Operating Handbook

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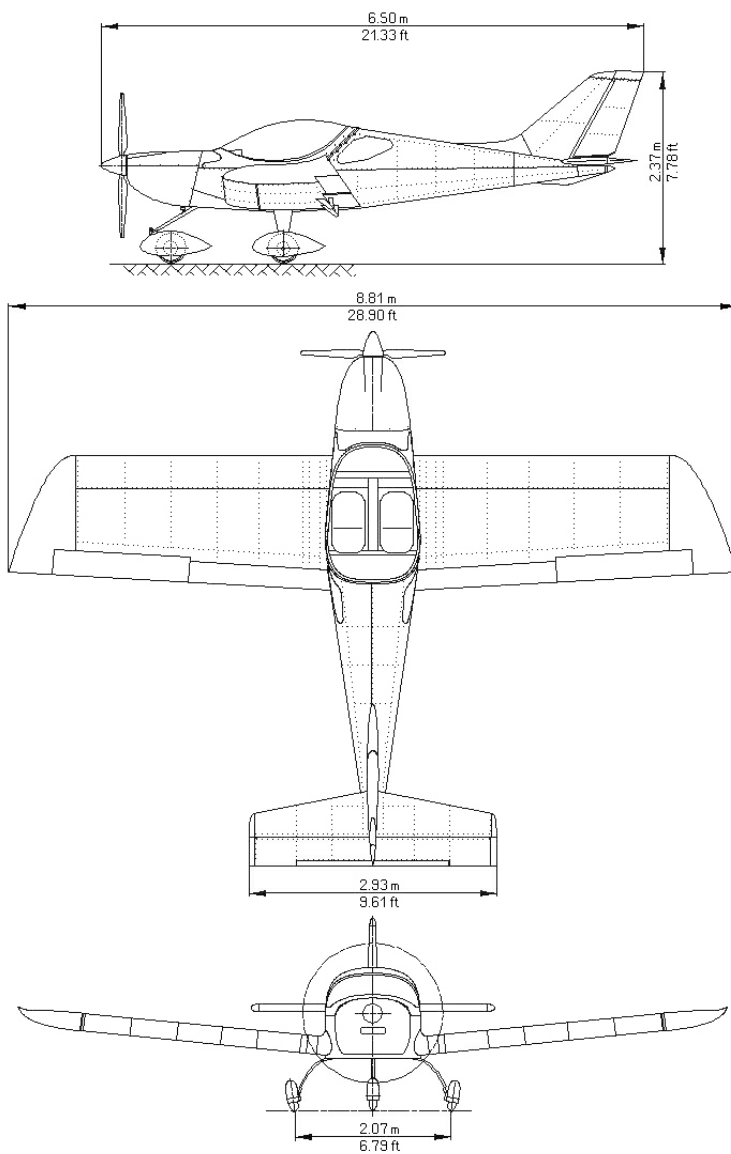
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Aircraft layout:



Main airplane dimensions:

Wing span.....	8.81 [m]	(28.90 [ft])
Length.....	6.50 [m]	(21.33 [ft])
Height	2.37 [m]	(7.78 [ft])
Wing area	12.3 [m ²]	(132.3 [sq ft])
Wing loading.....	49 [kg/m ²]	(10 [lb/sq ft])
Cockpit width	1.17 [m]	(46 [in])

Flight control surfaces travel:

Rudder.....	30° to each side
Elevator	+ 28°/- 25°
Aileron	+ 20°/- 15°
Flaps.....	0° to 30°
Aileron trim	+ 20°/- 20°
Elevator trim	+ 22°/- 28°

Engine:

Manufacturer	BRP-Rotax GmbH&Co.KG
Model number.....	912ULS
Maximum horsepower rating.....	73.5 kW (98.6 hp) at 5800 RPM
Cooling	liquid and air
Type	4-stroke, 4 cylinder, horizontally opposed, spark ignition engine with one central camshaft-push-rod-OHV

Propeller:

Manufacturer	WOODCOMP s.r.o.
Model number.....	KLASSIC 170/3/R
Number of blades	3
Diameter.....	1720 [mm] (68 [in])
Pitch setting.....	17.5 ± 0.5°
Type	three composite blades, ground adjustable

1.2 Summary of performances

Weights:

Max. takeoff and landing weight	560 [kg]	(1,235 [lb])
Max. weight of fuel.....	82 [kg]	(180 [lb])
Max. baggage weight in rear fuselage	18 [kg]	(40 [lb])
Max. baggage weight in each wing locker	20 [kg]	(44 [lb])
Empty weight (minimum equipment).....	345 [kg]	(760 [lb])

NOTE

Actual empty weight is shown in Section 9, Supplement No. 02

Wing loading.....	49 [kg/m ²]	(10 [lb/sq ft])
Power loading.....	8.15 [kg/kW]	(13.39 [lb/hp])

Speeds:

Maximum at sea level.....	116 [knot]	(215 [km/h])
Cruise, 75% power at 3,000 ft	93 [knot]	(172 [km/h])

Range and endurance:

Range.....	516 [NM]	(953 [km])
Endurance	5:25 [h:mm]	

Conditions:

Usable fuel.....	113 [liter]	(29.85 [US gal])
75% power of engine.....	5,000 RPM	
Altitude.....	3,000 [ft]	
Reserve	30 minutes	

Rate of climb:

At sea level.....	825 [fpm]	(4.2 [m/s])
Best angle of climb speed (v_x).....	56 [knot]	(104 [km/h])
Best rate of climb speed (v_y).....	62 [knot]	(115 [km/h])

Stall speeds:

V_{S0} – flaps down, power - idle.....	32 [knot]	(59 [km/h])
V_{S1} – flaps up, power - idle.....	39 [knot]	(72 [km/h])

Fuel:

Total fuel capacity.....	114 [liter]	(30.12 [US gal])
Total usable fuel.....	113 [liter]	(29.86 [US gal])
Approved types of fuel.....	see chapter 2.11	

Engine power:

Maximum power at 5,800 RPM.....	73.5 [kW]	(98.6 [hp])
Max. continuous power at 5,500 RPM.....	69 [kW]	(92.5 [hp])

2.3 Flap extended speed range - V_{S0} to V_{FE}

Flap operating range (IAS):

32 - 75 [knot] (59 - 139 [km/h])

2.4 Maneuvering speed - V_A

Maneuvering speed (IAS) at 600 [kg] (1,320 [lb]):

88 [knot] (163 [km/h])

Maneuvering speed (IAS) at 408 [kg] (900 [lb]):

70 [knot] (130 [km/h])

2.5 Maximum structural cruising speed – V_{NO}

Maximum structural cruising speed (IAS):

108 [knot] (200 [km/h])

2.6 Never exceed speed - V_{NE}

Never exceed speed (IAS):

138 [knot] (255 [km/h])

2.7 Service ceiling

Service ceiling 14,378 [ft] (4,382 [m])

2.8 Load factors

Maximum positive limit load factor + 4 g

Maximum negative limit load factor - 2 g

2.9 Approved maneuvers

The *PiperSport* is approved for normal and below listed maneuvers:

- Steep turns not exceeding 60° bank
- Lazy eights
- Chandelles
- Stalls (except whip stalls)

2.10 Operating weights and loading

Max. takeoff weight.....	560 [kg]	(1,235 [lb])
Max landing weight.....	560 [kg]	(1,235 [lb])
Max. weight of fuel.....	82 [kg]	(180 [lb])
Max. baggage weight in rear fuselage.....	18 [kg]	(40 [lb])
Max. baggage weight in each wing locker	20 [kg]	(44 [lb])
Empty weight (minimum equipment).....	345 [kg]	(760 [lb])

NOTE

Actual empty weight is shown in Section 9, Supplement No. 02

WARNING

Do not exceed maximum takeoff weight 560 [kg] (1,235 [lb]) !

Number of seats	2	
Minimum crew	1 pilot on the left seat	
Minimum crew weight	43 [kg]	(95 [lb])
Maximum crew weight on each seat.....	115 [kg]	(253 [lb])

2.11 Fuel

Fuel volume:

Wing fuel tanks capacity.....	2x57 [liter]	(2x15.06 [US gal])
Total fuel capacity.....	114 [liter]	(30.12 [US gal])
Unusable fuel.....	2x0.5 [liter]	(2x0.13 [US gal])
Total usable fuel	113 [liter]	(29.86 [US gal])

Recommended fuel type:

*(Refer to the ROTAX Operator's manual section 10.2.2 Fuel,
Rotax Service Instruction SI-912-016)*

MOGAS

European standard	- min. RON 95, EN 228 Super, EN 228 Super plus
US standard	- ASTM D4814
Canadian standard	- min. AKI 91, CAN/CGSB-3.5 Quality 3

CAUTION

Fuels that contain more than 5% ethanol blend have not been tested and are not permitted for use.

AVGAS

US standard	- AVGAS 100 LL (ASTM D910)
-------------	----------------------------

AVGAS 100 LL places greater stress on the valve seats due to its high lead content and forms increased deposits in the combustion chamber and lead sediments in the oil system. Thus it should only be used in case of problems with vapor lock or when other types of gasoline are unavailable.

2.12 Engine operating speeds and limits

Engine Model:		ROTAX 912 ULS
Engine Manufacturer:		Bombardier-Rotax GMBH
Power	Max. Takeoff:	73.5 kW (98.6 hp) at 5,800 rpm (max. 5 min.)
	Max. Continuous:	69 kW (92.5 hp) at 5,500 rpm
	Cruising (75%):	51 kW (68.4 hp) at 5,000 rpm
Engine RPM	Max. Takeoff:	5,800 rpm (max. 5 min)
	Max. Continuous:	5,500 rpm
	Cruising (75%):	5,000 rpm
	Idling:	1,400 rpm (minimum)
Oil pressure	Minimum:	0.8 bar (12 psi) below 3,500 rpm
	Maximum:	7 bar (102 psi) cold engine starting
	Optimum:	2 - 5 bar (29 - 73 psi) above 3,500 rpm
Oil temperature	Minimum:	50° C (122° F)
	Maximum:	130° C (266° F)
	Optimum:	90 - 110° C (194 - 230° F)
Cylinder head temper. (CHT)	Minimum:	50° C (122° F)
	Maximum:	135° C (275° F) *
Exhaust gas temperature (EGT)	Nominal:	800° C (1472° F)
	Maximum:	850° C (1562° F)
	Max. Takeoff:	880° C (1616° F)
Fuel press.	Minimum:	0.15 bar (2.2 psi)
	Maximum:	0.4 bar (5.8 psi)

* See the Rotax Operator's manual section 10.1.2 Operating speeds and limits and section 10.2.1 Coolant, Rotax Installation manual section 12 Cooling system, Rotax Service Instruction SI-912-016, POH Section 8, paragraph 8.5.3 Coolant and Section 9, Supplement No. 02. Type of coolant used in engine.

2.13 Engine instruments markings

Rotax 912ULS 73.5 [kW] (98.6 [hp])	Minimum Limit (red line)	Caution Range (yellow arc)	Normal Operating Range (green arc)	Caution Range (yellow arc)	Maximum Range (red line)
Engine speed [RPM]	-	0-1400	1400-5500	5500-5800	5800
Oil Pressure	0.8 bar (12 psi)	0.8-2 bar (12-29 psi)	2-5 bar (29-73 psi)	5-7 bar (73-102 psi)	7 bar (102 psi)
Oil Temperature	50°C (122°F)	50-90°C (122-194°F)	90-110°C (194-230°F)	110-130°C (230-266°F)	130°C (266°F)
Cylinder head Temperature (CHT)	-	-	50-135°C (122-275°F)	-	135°C (275°F)
Exhaust Gas Temp. (EGT)	-	400-500°C (752-932°F)	500-850°C (932-1562°F)	850-880°C (1562-1616°F)	880°C (1616°F)
Fuel Pressure	0.15 bar (2.2 psi)	-	0.15-0.4 bar (2.2-5.8 psi)	-	0.4 bar (5.8 psi)
Manifold Pressure	-	-	10-29 inHg	-	29 inHg

2.14 Other limitations

- **No smoking on board of the aircraft!**
- **Approved for Day VFR flights only.**
- **Flight in rain**

When flying in the rain, no additional steps are required. Aircraft qualities and performance are not substantially changed. However **VMC must be maintained !**

• ***Minimum instruments and equipment list for Day VFR flights:***

- Airspeed indicator
- Altimeter
- Compass (is not required by ASTM F 2245)
- Fuel quantity indicator
- Tachometer (RPM)
- Engine instruments as required by the engine manufacturer :
 - *Oil temperature indicator*
 - *Oil pressure indicator*
 - *Cylinder head temperature indicator*
- Safety harness for every used seat

WARNING

IFR FLIGHTS AND INTENTIONAL FLIGHTS UNDER ICING CONDITIONS ARE PROHIBITED!

WARNING

EMERGENCY PARACHUTE APPROVED FOR UP TO MTOW 612 KG (1,350 LBS) AND MAX. VELOCITY 120 KNOTS (222 KM/H)!

WARNING

MINIMUM 6 LITRES (1.58 US GAL) OF FUEL QUANTITY ALLOWS APPROXIMATELY 15 MINUTES OF SAFE OPERATION!

CAUTION

Install air intake shields in front of water and oil cooler, if ambient air temperature is 0°C (32°F) or lower.

CAUTION

- *The starter should be activated for a maximum of 10 [sec], followed by 2 [min] pause for starter cooling.*
- *As soon as engine runs, adjust throttle to achieve smooth running at approx. 2,500 [rpm]. Check the oil pressure, which should increase within 10 [sec]. Increase the engine speed after the oil pressure has reached 2 [bar] (29 [psi]) and is steady.*
- *To avoid shock loading, start the engine with the throttle lever set for idling or 10 % open at maximum, then wait 3 [sec] to reach constant engine speed before new acceleration.*
- *Only one magneto should be switched ON (OFF) during ignition magneto check.*

4.2.3 Engine warm up, Engine check

Prior to engine check block the main wheels using chocks. Initially warm up the engine to 2,000 [rpm] for approximately 2 [min], then continue to 2,500 [rpm] till oil temperature reaches 50 [°C] (122 [°F]). The warm up period depends on ambient air temperature.

Check both ignition circuits at 4,000 [rpm] for Rotax 912 ULS. The engine speed drop during the time either magneto switched OFF should not exceed 300 [rpm]. The max. engine speed drop difference between circuits L and R should be 115 [rpm].

NOTE

Only one magneto should be switched ON (OFF) during ignition magneto check.

Set max. power for verification of max. static engine speed (5,000 ±100 [rpm]) with given propeller and engine parameters (*temperatures and pressures*).

Check acceleration from idling to max. power. If necessary, cool the engine at *idle [rpm]* before shutdown.

CAUTION

The engine check should be performed with the aircraft heading upwind and not on a loose terrain (the propeller may suck grit which can damage the leading edges of blades).

4.3 Taxiing

Apply power and brakes as needed. Apply brakes to control movement on ground. Taxi carefully when wind velocity exceeds 20 [knot] (11 [m/s]). Hold the control stick in neutral position.

4.4 Normal Takeoff

4.4.1 Before takeoff

1. Altimeter - set
2. Trim - set neutral position
3. Control system - check free movement
4. Cockpit canopy - closed and locked

Recommendation: - **Before takeoff, manually check the canopy is locked by pushing the canopy upwards.**

5. Safety harness - fastened
6. Fuel selector - turn ON (*left or right fuel tank*)
7. Ignition switch - switched ON (*both magnetos*)
8. Wing flaps - extend as necessary

4.4.2 Takeoff

1. Brakes - apply to stop wheel rotation
2. Throttle - fully forward
3. Engine speed - check (*5,000 ±100 [rpm]*)
4. Engine gauges - within limits
5. Brakes - release
6. Elevator - control stick pull
7. Nose wheel unstick - *32 [knot] (59 [km/h])*
8. Airplane lift-off - *42 [knot] (78 [km/h])*
9. Climb - after reaching airspeed
65 [knot] (120 [km/h])
10. Wing flaps - retract at safe altitude
*(max. airspeed for flaps using is
75 [knot], 139 [km/h])*

WARNING

Takeoff is prohibited if:

- Engine is running unsteadily
- Engine instrument values are beyond operational limits
- Aircraft systems (e.g. brakes or controls) work incorrectly
- Crosswind velocity exceeds permitted limits
(see Section 5 Performance, 5.7 Demonstrated wind performance)

4.5 Climb

1. Throttle - fully forward
(max. 5,800 [rpm] for max. 5 [min],
max. continuous power (5,500 [rpm]))
2. Airspeed - $V_x = 56$ [knot] (104 [km/h])
- $V_y = 62$ [knot] (115 [km/h])
3. Trim - trim the airplane
4. Engine gauges - oil temperature, oil pressure and
CHT within limits

CAUTION

If the cylinder head temperature or oil temperature and/or coolant temperature approaches or exceeds limits, reduce the climb angle to increase airspeed and possibly return within limits. If readings do not improve, troubleshoot causes other than high power setting at low airspeed.

4.6 Best angle of climb speed (V_x): 56 [knot] (104 [km/h])

4.7 Best rate of climb speed (V_y): 62 [knot] (115 [km/h])

4.8 Cruise

Refer to Section 5, for recommended cruising figures

4.9 Descend

1. Optimum glide speed - 60 [knot] (110 [km/h])

4.10 Approach

1. Approach speed - 60 [knot] (110 [km/h])
2. Throttle - as necessary
3. Wing flaps - extend as necessary
4. Trim - as necessary
5. Safety harness - tighten

CAUTION

It is not advisable to reduce the engine throttle control lever to minimum on final approach and when descending from very high altitude. In such cases the engine becomes under-cooled and a loss of power may occur. Descent at increased idle (approximately 3,000 [rpm]), speed between 60-75 [knot] (110-139 [km/h]) and check that the engine instruments indicate values within permitted limits.

4.11 Normal landing

4.11.1 Before landing

1. Throttle - as necessary
2. Airspeed - 60 [knot] (110 [km/h])
3. Wing flaps - extend as necessary
4. Trim - as necessary

4.11.2 Landing

1. Throttle - idle
2. Touch-down on main wheels
3. Apply brakes - as necessary
(after the nose wheel touch-down)

4.11.3 After landing

1. Throttle - engine rpm set as required for taxiing
2. Wing flaps - retract
3. Trim - set neutral position

4.11.4 Engine shut down

- | | |
|----------------------|---|
| 1. Throttle | - idle |
| 2. Instruments | - engine instruments within limits |
| 3. Switches | - switch OFF - except Instrument and Master |
| 4. Ignition switch | - turn key to switch OFF |
| 5. Instrument switch | - switch OFF |
| 6. Master switch | - switch OFF |
| 7. Fuel Selector | - close |

CAUTION

Rapid engine cooling should be avoided during operation. This happens above all during aircraft descent, taxiing and low engine [rpm] or at engine shutdown immediately after landing. Under normal conditions the engine temperatures stabilize during descent, taxiing and at values suitable to stop engine by switching the ignition off. If necessary, cool the engine at *idle* [rpm] to stabilize the temperatures prior to engine shut down.

4.12 Short field takeoff and landing procedures

None

4.13 Balked landing procedures

- | | |
|--------------------------|--|
| 1. Throttle | - fully forward
(max. 5,800 [rpm] for max. 5 [min],
max. continuous power (5,500 [rpm])) |
| 2. Climb | - after reaching
65 [knot] (120 [km/h]) |
| 3. Trim | - adjust as necessary |
| 4. Wing flaps | - retract at safe altitude
(max. airspeed for flaps using is
75 [knot], 139 [km/h]) |
| 5. Trim | - adjust as necessary |
| 6. Repeat circle pattern | |

4.14 Aircraft parking and tie-down

1. Ignition switch - OFF
2. Master switch - OFF
3. Fuel selector - close
4. Parking brake - use it as necessary (if installed)
5. Canopy - close, lock as necessary
6. Secure the airplane

NOTE

It is recommended to use parking brake (if installed) for short-time parking only, between flights during a flight day. After ending the flight day or at low temperatures of ambient air, do not use parking brake, but use the wheel chocks instead.

NOTE

Use anchor eyes on the wings and fuselage rear section to fix the airplane. Move control stick forward and fix it together with the rudder pedals. Make sure that the cockpit canopy is properly closed and locked. The anchoring before leaving the airplane is important if the airplane is not equipped with a parking brake.

SECTION 5

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5. PERFORMANCE

The presented data has been computed from actual flight tests with the aircraft and engine in good conditions and using average piloting techniques. If not stated otherwise, the performance stated in this section is valid for maximum take-off weight (600 [kg]/1,320 [lb]) and under ISA conditions.

The performance shown in this section is valid for aircraft equipped with **ROTAX 912 ULS** engine with maximum power 73.5 [kW] (98.6 [hp]) and **WOODCOMP KLASSIC 170/3/R** three composite blades ground adjustable propeller with pitch setting 17.5°.

CAUTION

Airspeeds values are valid for standard AVIATIK WA037383 pitot-static probe.

5.1 Takeoff distances

- Conditions: - Altitude: 0 ft ISA
 - Engine power: full throttle
 - Flaps: 15°

RUNWAY SURFACE	Take-off run distance		Take-off distance over 50 ft (15 m) obstacle	
	ft	m	ft	m
CONCRETE	463	141	1,270	387
GRASS	702	214	1,499	457

5.2 Landing distances

- Conditions: - Altitude: 0 ft ISA
 - Engine power: idle
 - Flaps: 30°
 - Brakes fully depressed immediately after touch-down

RUNWAY SURFACE	Landing distance over 50 ft (15 m) obstacle		Landing run distance (braked)	
	ft	m	ft	m
CONCRETE	1,188	362	479	146
GRASS	1,109	338	364	111

5.3 Rate of climb

Conditions: Engine: full throttle Flaps: 0°	Best rate of climb speed Vy (IAS)		Rate of climb Vz	
	knot	km/h	fpm	m/s
Altitude				
0 ft	62	115	825	4.2
3000 ft	62	115	685	3.5
6000 ft	62	115	540	2.7
9000 ft	62	115	355	1.8

5.4 Cruise speeds

Altitude	Engine speed	IAS		CAS		TAS		MAP	Fuel consumption	
		knot	km/h	knot	km/h	knot	km/h		in Hg	l/h
1,000	4,200	72	133	72	133	73	135	23.7	13.6	3.59
	4,500	81	150	80	148	81	150	24.6	15.7	4.15
	4,800	91	169	89	165	89	165	25.5	18.0	4.76
	5,000	96	178	94	174	95	176	26.1	19.5	5.15
	5,300	105	194	102	189	103	191	27.0	21.9	5.79
	5,500	112	207	108	200	109	202	27.7	23.7	6.26
	5,700	118	219	113	209	114	211	28.3	25.8	6.82
3,000	4,200	68	126	69	128	72	133	22.2	13.2	3.49
	4,500	78	144	77	143	80	148	23.0	15.3	4.04
	4,800	86	159	85	157	88	163	23.8	17.5	4.62
	5,000	93	172	91	169	94	174	24.3	19.0	5.02
	5,300	102	189	99	183	102	189	25.1	21.4	5.65
	5,500	108	200	104	193	108	200	25.5	23.3	6.16
5,000	4,200	65	120	66	122	71	131	20.5	12.9	3.41
	4,500	74	137	74	137	79	146	21.3	14.9	3.94
	4,800	83	154	82	152	87	161	22.1	17.2	4.54
	5,000	89	165	87	161	93	172	22.7	18.7	4.94
	5,300	97	180	95	176	101	187	23.5	21.1	5.57
	5,500	103	191	100	185	107	198	24.1	22.8	6.02
7,000	4,200	62	115	63	117	69	128	19.3	12.5	3.30
	4,500	69	128	70	130	77	143	20.0	14.6	3.86
	4,800	79	146	78	144	85	157	20.6	16.8	4.44
	5,000	84	156	83	154	91	169	21.2	18.4	4.86
	5,300	92	170	90	167	99	183	22.0	20.8	5.49
	5,500	98	181	95	176	105	194	22.5	22.3	5.89
9,000	4,200	57	106	59	109	67	124	18.4	12.2	3.22
	4,500	64	119	65	120	74	137	19.0	14.3	3.86
	4,800	73	135	73	135	83	154	19.6	16.4	4.33
	5,000	79	146	78	144	89	165	20.0	18.0	4.76
	5,300	86	159	85	157	97	180	20.5	20.4	5.39
	5,500	92	170	90	167	103	191	20.8	21.8	5.76

5.5 RPM setting and fuel consumption

Altitude		ft	1,000					
Engine speed		<i>rpm</i>	4,200	4,500	4,800	5,000	5,300	5,500
Fuel consumption		<i>l/h</i>	13.6	15.7	18.0	19.5	21.9	23.7
		<i>US gal/h</i>	3.59	4.15	4.76	5.15	5.79	6.26
Airspeed	IAS	<i>knot</i>	72	81	91	96	105	112
		<i>km/h</i>	133	150	169	178	194	207
	CAS	<i>knot</i>	72	80	89	94	102	108
		<i>km/h</i>	133	148	165	174	189	200
	TAS	<i>knot</i>	73	81	89	95	103	109
		<i>km/h</i>	135	150	165	176	191	202
Endurance and Range at 113 liters (29.85 US gal)								
Endurance		<i>hh:mm</i>	8:18	7:11	6:16	5:47	5:09	4:46
Range		<i>NM</i>	607	583	559	551	531	520
		<i>km</i>	1123	1080	1035	1020	984	962
Endurance and Range at 90 liters (23.77 US gal)								
Endurance		<i>hh:mm</i>	6:37	5:43	5:00	4:36	4:06	3:47
Range		<i>NM</i>	483	464	445	438	423	414
		<i>km</i>	895	860	824	812	784	767
Endurance and Range at 60 liters (15.85 US gal)								
Endurance		<i>hh:mm</i>	4:24	3:49	3:20	3:04	2:44	2:31
Range		<i>NM</i>	322	310	297	292	282	276
		<i>km</i>	596	573	549	541	523	511
Endurance and Range at 30 liters (7.92 US gal)								
Endurance		<i>hh:mm</i>	2:12	1:54	1:40	1:32	1:22	1:15
Range		<i>NM</i>	161	155	148	146	141	138
		<i>km</i>	298	287	275	271	261	256
Endurance and Range at 15 liters (3.96 US gal)								
Endurance		<i>hh:mm</i>	1:06	0:57	0:50	0:46	0:41	0:37
Range		<i>NM</i>	81	77	74	73	71	69
		<i>km</i>	149	143	137	135	131	128

Altitude		<i>ft</i>	3,000					
Engine speed	<i>rpm</i>		4,200	4,500	4,800	5,000	5,300	5,500
Fuel consumption	<i>l/h</i>		13.2	15.3	17.5	19.0	21.4	23.3
	<i>US gal/h</i>		3.49	4.04	4.62	5.02	5.65	6.16
Airspeed	IAS	<i>knot</i>	68	78	86	93	102	108
		<i>km/h</i>	126	144	159	172	189	200
	CAS	<i>knot</i>	69	77	85	91	99	104
		<i>km/h</i>	128	143	157	169	183	193
	TAS	<i>knot</i>	72	80	89	95	103	109
		<i>km/h</i>	133	148	165	176	191	202
Endurance and Range at 113 liters (29.85 US gal)								
Endurance	<i>hh:mm</i>		8:33	7:23	6:27	5:56	5:16	4:50
Range	<i>NM</i>		616	591	568	559	539	524
	<i>km</i>		1142	1094	1052	1035	997	970
Endurance and Range at 90 liters (23.77 US gal)								
Endurance	<i>hh:mm</i>		6:49	5:52	5:08	4:44	4:12	3:51
Range	<i>NM</i>		491	471	453	445	429	417
	<i>km</i>		909	872	838	825	794	773
Endurance and Range at 60 liters (15.85 US gal)								
Endurance	<i>hh:mm</i>		4:32	3:55	3:25	3:09	2:48	2:34
Range	<i>NM</i>		327	314	302	297	286	278
	<i>km</i>		606	581	559	550	530	515
Endurance and Range at 30 liters (7.92 US gal)								
Endurance	<i>hh:mm</i>		2:16	1:57	1:42	1:34	1:24	1:17
Range	<i>NM</i>		164	157	151	148	143	139
	<i>km</i>		303	291	279	275	265	258
Endurance and Range at 15 liters (3.96 US gal)								
Endurance	<i>hh:mm</i>		1:08	0:58	0:51	0:47	0:42	0:38
Range	<i>NM</i>		82	78	75	74	71	70
	<i>km</i>		152	145	140	137	132	129

Altitude		ft	5,000					
Engine speed		<i>rpm</i>	4,200	4,500	4,800	5,000	5,300	5,500
Fuel consumption		<i>l/h</i>	12.9	14.9	17.2	18.7	21.1	22.8
		<i>US gal/h</i>	3.41	3.94	4.54	4.94	5.57	6.02
Airspeed	IAS	<i>knot</i>	65	74	83	89	97	103
		<i>km/h</i>	120	137	154	165	180	191
	CAS	<i>knot</i>	66	74	82	87	95	100
		<i>km/h</i>	122	137	152	161	176	185
	TAS	<i>knot</i>	71	79	87	93	101	107
		<i>km/h</i>	131	146	161	172	187	198
Endurance and Range at 113 liters (29.85 US gal)								
Endurance		<i>hh:mm</i>	8:45	7:35	6:34	6:02	5:21	4:57
Range		<i>NM</i>	622	599	572	562	541	530
		<i>km</i>	1152	1110	1059	1041	1002	982
Endurance and Range at 90 liters (23.77 US gal)								
Endurance		<i>hh:mm</i>	6:58	6:02	5:13	4:48	4:15	3:56
Range		<i>NM</i>	495	477	455	448	431	422
		<i>km</i>	917	884	843	829	798	782
Endurance and Range at 60 liters (15.85 US gal)								
Endurance		<i>hh:mm</i>	4:39	4:01	3:29	3:12	2:50	2:37
Range		<i>NM</i>	330	318	303	298	287	282
		<i>km</i>	612	589	562	553	532	521
Endurance and Range at 30 liters (7.92 US gal)								
Endurance		<i>hh:mm</i>	2:19	2:00	1:44	1:36	1:25	1:18
Range		<i>NM</i>	165	159	152	149	144	141
		<i>km</i>	306	295	281	276	266	261
Endurance and Range at 15 liters (3.96 US gal)								
Endurance		<i>hh:mm</i>	1:09	1:00	0:52	0:48	0:42	0:39
Range		<i>NM</i>	83	80	76	75	72	70
		<i>km</i>	153	147	141	138	133	130

Altitude		<i>ft</i>	7,000					
Engine speed		<i>rpm</i>	4,200	4,500	4,800	5,000	5,300	5,500
Fuel consumption		<i>l/h</i>	12.5	14.6	16.8	18.4	20.8	22.3
		<i>US gal/h</i>	3.30	3.86	4.44	4.86	5.49	5.89
Airspeed	IAS	<i>knot</i>	62	69	79	84	92	98
		<i>km/h</i>	115	128	146	156	170	181
	CAS	<i>knot</i>	63	70	78	83	90	95
		<i>km/h</i>	117	130	144	154	167	176
	TAS	<i>knot</i>	69	77	85	91	99	105
		<i>km/h</i>	128	143	157	169	183	194
Endurance and Range at 113 liters (29.85 US gal)								
Endurance		<i>hh:mm</i>	9:02	7:44	6:43	6:08	5:25	5:04
Range		<i>NM</i>	624	596	572	559	538	532
		<i>km</i>	1155	1104	1059	1035	996	985
Endurance and Range at 90 liters (23.77 US gal)								
Endurance		<i>hh:mm</i>	7:12	6:09	5:21	4:53	4:19	4:02
Range		<i>NM</i>	497	475	455	445	428	424
		<i>km</i>	920	879	843	824	793	785
Endurance and Range at 60 liters (15.85 US gal)								
Endurance		<i>hh:mm</i>	4:48	4:06	3:34	3:15	2:53	2:41
Range		<i>NM</i>	331	316	304	297	286	283
		<i>km</i>	613	586	562	550	529	523
Endurance and Range at 30 liters (7.92 US gal)								
Endurance		<i>hh:mm</i>	2:24	2:03	1:47	1:37	1:26	1:20
Range		<i>NM</i>	166	158	152	148	143	141
		<i>km</i>	307	293	281	275	264	262
Endurance and Range at 15 liters (3.96 US gal)								
Endurance		<i>hh:mm</i>	1:12	1:01	0:53	0:48	0:43	0:40
Range		<i>NM</i>	83	79	76	74	71	71
		<i>km</i>	153	147	141	137	132	131

Altitude		<i>ft</i>	9,000					
Engine speed	<i>rpm</i>		4,200	4,500	4,800	5,000	5,300	5,500
Fuel consumption	<i>l/h</i>		12.2	14.3	16.4	18.0	20.4	21.8
	<i>US gal/h</i>		3.22	3.78	4.33	4.76	5.39	5.76
Airspeed	IAS	<i>knot</i>	57	64	73	79	86	92
		<i>km/h</i>	106	119	135	146	159	170
	CAS	<i>knot</i>	59	65	73	78	85	90
		<i>km/h</i>	109	120	135	144	157	167
	TAS	<i>knot</i>	67	74	83	89	97	103
		<i>km/h</i>	124	137	154	165	180	191
Endurance and Range at 113 liters (29.85 US gal)								
Endurance	<i>hh:mm</i>		9:15	7:54	6:53	6:16	5:32	5:11
Range	<i>NM</i>		621	585	572	559	537	534
	<i>km</i>		1149	1083	1059	1035	995	989
Endurance and Range at 90 liters (23.77 US gal)								
Endurance	<i>hh:mm</i>		7:22	6:17	5:29	5:00	4:24	4:07
Range	<i>NM</i>		494	466	455	445	428	425
	<i>km</i>		915	863	844	824	793	788
Endurance and Range at 60 liters (15.85 US gal)								
Endurance	<i>hh:mm</i>		4:55	4:11	3:39	3:20	2:56	2:45
Range	<i>NM</i>		330	310	304	297	285	283
	<i>km</i>		610	575	562	549	528	525
Endurance and Range at 30 liters (7.92 US gal)								
Endurance	<i>hh:mm</i>		2:27	2:05	1:49	1:40	1:28	1:22
Range	<i>NM</i>		165	155	152	148	143	142
	<i>km</i>		305	288	281	275	264	263
Endurance and Range at 15 liters (3.96 US gal)								
Endurance	<i>hh:mm</i>		1:13	1:02	0:54	0:50	0:44	0:41
Range	<i>NM</i>		82	78	76	74	71	71
	<i>km</i>		153	144	141	137	132	131

5.6 Airspeed indicator system calibration

IAS	CAS
<i>knot</i>	
30	35
35	39
40	44
45	48
50	53
55	57
60	62
65	66
70	71
75	75
80	79
85	84
90	88
95	93
100	97
105	102
110	106
115	111
120	115
125	120
130	124
135	129
140	133

IAS	CAS
<i>km/h</i>	
55	64
60	68
70	77
80	86
90	95
100	104
110	113
120	122
130	131
140	140
150	149
160	158
170	167
180	176
190	185
200	194
210	203
220	212
230	221
240	230
250	239
255	243

5.7 Demonstrated wind performance

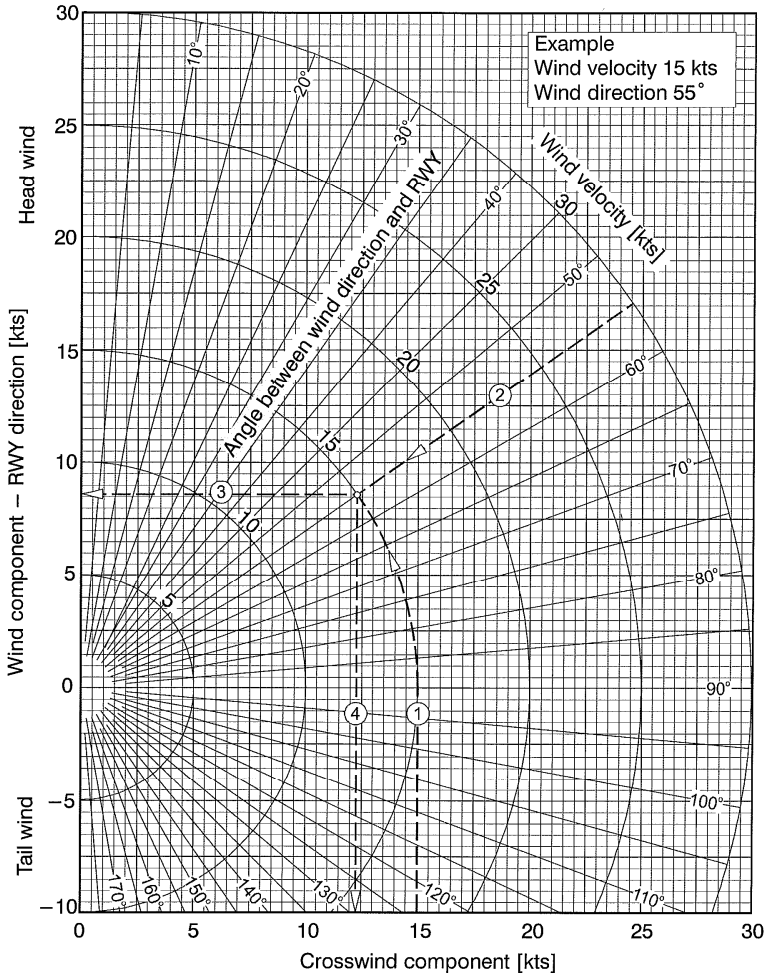
Max. demonstrated headwind velocity for take-off and landing:

24 [knot] (13 [m/s])

Max. demonstrated crosswind velocity for take-off and landing:

12 [knot] (6,5 [m/s])

Wind components figure



Example: 1. Wind velocity 15 knots 3. Headwind component 8.6 knots
 2. Wind direction 55° 4. Crosswind component 12.3 knots

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8.3 Towing instructions

To handle the airplane on ground use the *Tow Bar*, or if pushing the airplane by hand, push on the aft fuselage, placing your hands over an area of skin supported by a bulkhead.

CAUTION

*Avoid excessive pressure at the airplane airframe-especially at control surfaces.
Keep all safety precautions, especially in the propeller area.*

8.4 Tie-down instructions

The airplane should be moored when parked outside a hangar after the flight day. The mooring is necessary to protect the airplane against possible damage caused by wind and gusts.

For this reason the aircraft is equipped with mooring eyes located on the lower surfaces of the wings.

Tie-down procedures:

1. Fuel Selector close, Master switch and other switches switched OFF, Ignition switch switched OFF
2. Fix the hand control using e.g. safety harness
3. Close air vent
4. Close and lock canopy
5. Moor the aircraft to the ground by means of a mooring rope passed through the mooring eyes located on the lower surfaces of the wings and below rear fuselage.

NOTE

In the case of long term parking, especially during winter, it is recommended to cover the cockpit canopy or possibly the whole aircraft by means of a suitable tarpaulin attached to the airframe.

8.5 Servicing operating fluids

See appropriate chapters in the ROTAX engine Maintenance and Operator's manuals and *PiperSport* aircraft Maintenance manual for more instructions.

8.5.1 Approved fuel grades and specifications

Recommended fuel type:

(refer to the ROTAX Operator's manual section 10.2.2 Fuel,
Rotax Service Instruction SI-912-016)

MOGAS

- European standard - min. RON 95, EN 228 Super, EN 228 Super plus
- US standard - ASTM D4814
- Canadian standard - min. AKI 91, CAN/CGSB-3.5 Quality 3

CAUTION

Fuels that contain more than 5% ethanol blend have not been tested and are not permitted for use.

AVGAS

- US standard - AVGAS 100 LL (ASTM D910)

AVGAS 100 LL places greater stress on the valve seats due to its high lead content and forms increased deposits in the combustion chamber and lead sediments in the oil system. Thus it should only be used in case of problems with vapor lock or when other types of gasoline are unavailable.

Fuel volume:

- Wing fuel tanks volume.....2x57 [liter] (2x15.06 [US gal])
- Unusable fuel quantity2x0.5 [liter] (2x0.13 [US gal])

8.5.2 Approved oil grades and specifications

Recommended oil type:

(refer to the Rotax Operator's manual section 10.2.3 Lubricants,
Rotax Service Instruction SI-912-016)

Motorcycle 4-stroke engine oil of registered brand with gear additives.

Use only oil with API "SG" classification or higher!

Use multi-grade oil. Use of mineral oil is not recommended.

Type of oil used by aircrafts manufacturer:

- see Section 9, Supplement No. 02

Oil volume:

- Minimum3.3 [liter] (0.87 [US gal])
- Maximum3.8 [liter] (1.0 [US gal])